

Passive control of fluid-structure instabilities by means of piezo-shunts

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AEROFLEX project



return on innovation

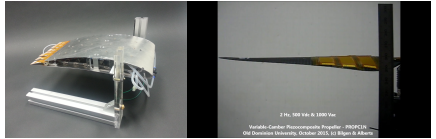
Piezo-electricity : a wide range of applications

Piezo-electricity : *Electric response of certain materials to a mechanical stress due to their microscopic structure*

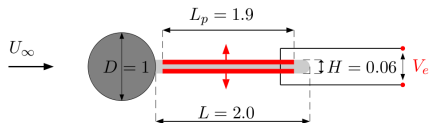
Sensing & Actuation from everyday life ...



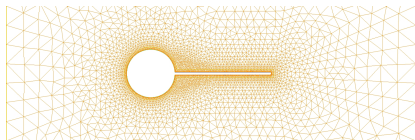
... to the **Aerospace Research**



Our Goal : using piezos to control fluid-structure instabilities



Fluid region



Solid region



- Rigid cylinder + elastic plate + piezo patches.
- $Re = U_\infty D / \nu = 80$, density ratio $m_\rho = 50$ and bending stiffness $K = 0.3$.
- Electro-mechanical coupling only acting for bending modes (*transverse effect + opposite poiling direction*).

Arbitrary Lagrangian Eulerian approach

- Lagrangian for the solid.
 - Eulerian description for the fluid.
 - Eqs. in the underformed configuration
- Fully coupled nonlinear **fluid-electro-mechanical** system in compact form

$$\mathcal{B} \frac{\partial \mathbf{q}}{\partial t} = \mathcal{R}(\mathbf{q}).$$

- Linearization around a fixed equilibrium solution \mathbf{q}_B , i.e. $\mathcal{R}(\mathbf{q}_B) = \mathbf{0}$, with $\mathbf{q}(\mathbf{x}, t) = \hat{\mathbf{q}}(\mathbf{x})e^{\lambda t}$

$$\lambda \mathcal{B} \hat{\mathbf{q}} = \left. \frac{\partial \mathcal{R}(\mathbf{q})}{\partial \mathbf{q}} \right|_{\mathbf{q}_B} \hat{\mathbf{q}},$$

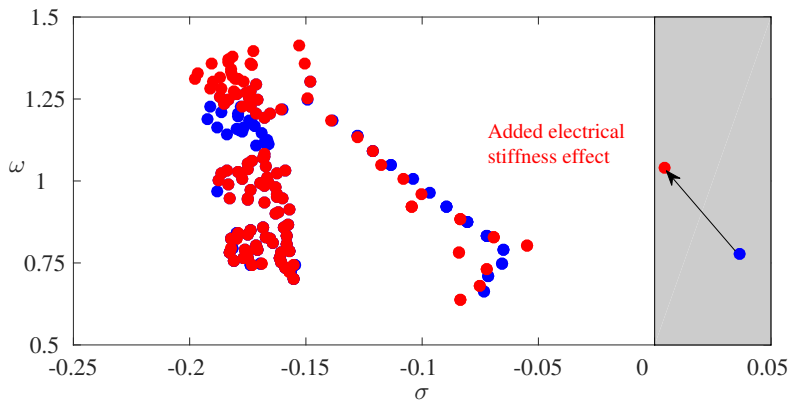
which is a **Generalised Eigenvalue Problem**, for $\lambda = \sigma + i\omega \in \mathbb{C}$:

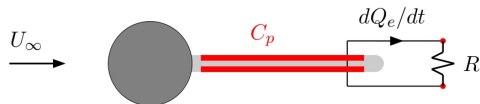
Base flow

No Piezos

Piezos

Without vs With Piezos



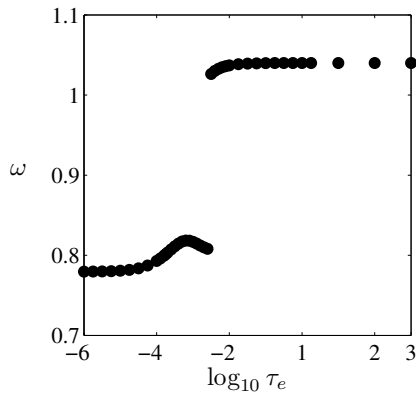
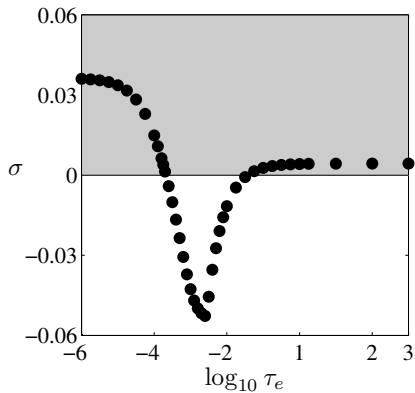


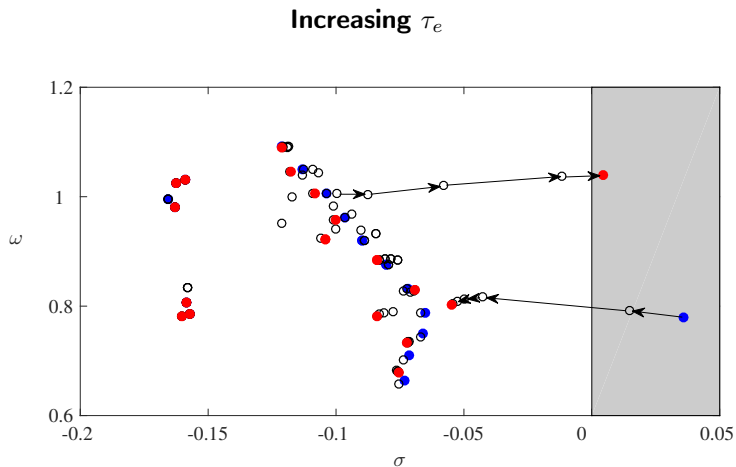
1st order electr. dynamics

$$\frac{dQ_e}{dt} + \frac{1}{RC_p} Q_e = 0,$$

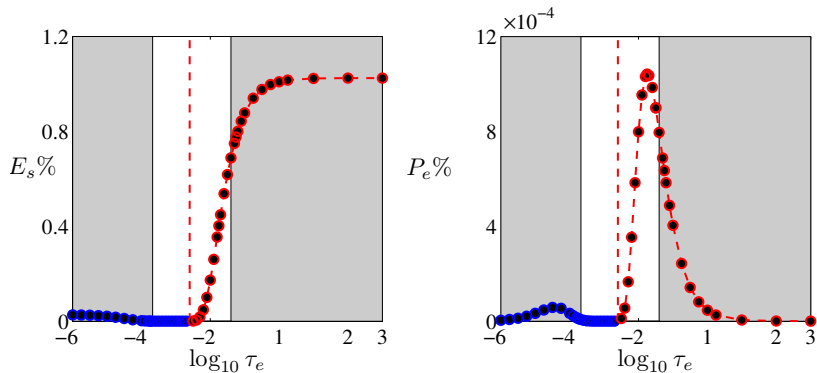
- $\tau_e = RC_p$ is the *characteristic electrical time*.
- **Energy dissipation** by Joule's effect, $P_e = C_p V_e^2 / \tau_e$.
- For $\tau_e \rightarrow 0$, **electro-mechanical coupling** becomes **negligible** (short-circuit).
- For $\tau_e \rightarrow \infty$, **electro-mechanical coupling** becomes **maximal** (open-circuit).

Leading eigenvalue





Leading mode



A *conjecture* on two possible distinct **stabilization mechanisms** :

- **Frequency desynchronization** by means of the added electric stiffness (*original flutter mode*).
- **Electrical damping** through Joule effect (*new selected flutter mode*).

Thanks for your attention.

(Any Questions?)